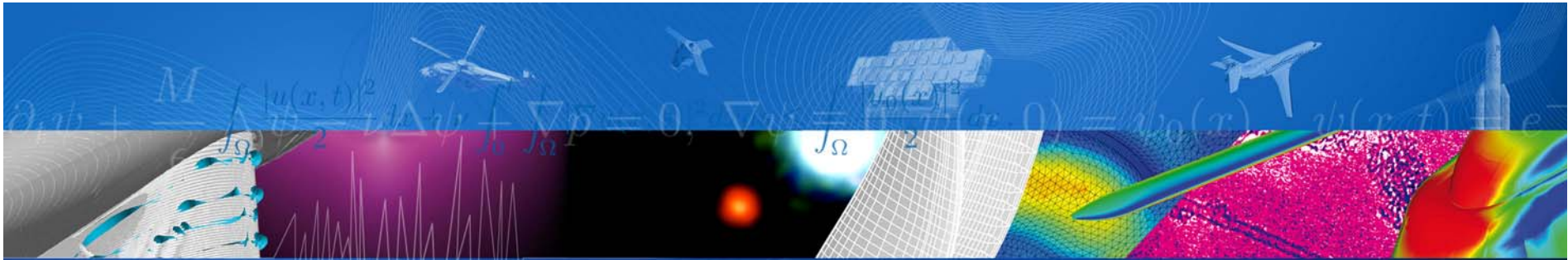


ONERA

THE FRENCH AEROSPACE LAB

retour sur innovation

[www.onera.fr](http://www.onera.fr)



# Significant achievements and future research in the field of environmental impact of aeronautics at Onera

**Nagoya - October 2012**



retour sur innovation





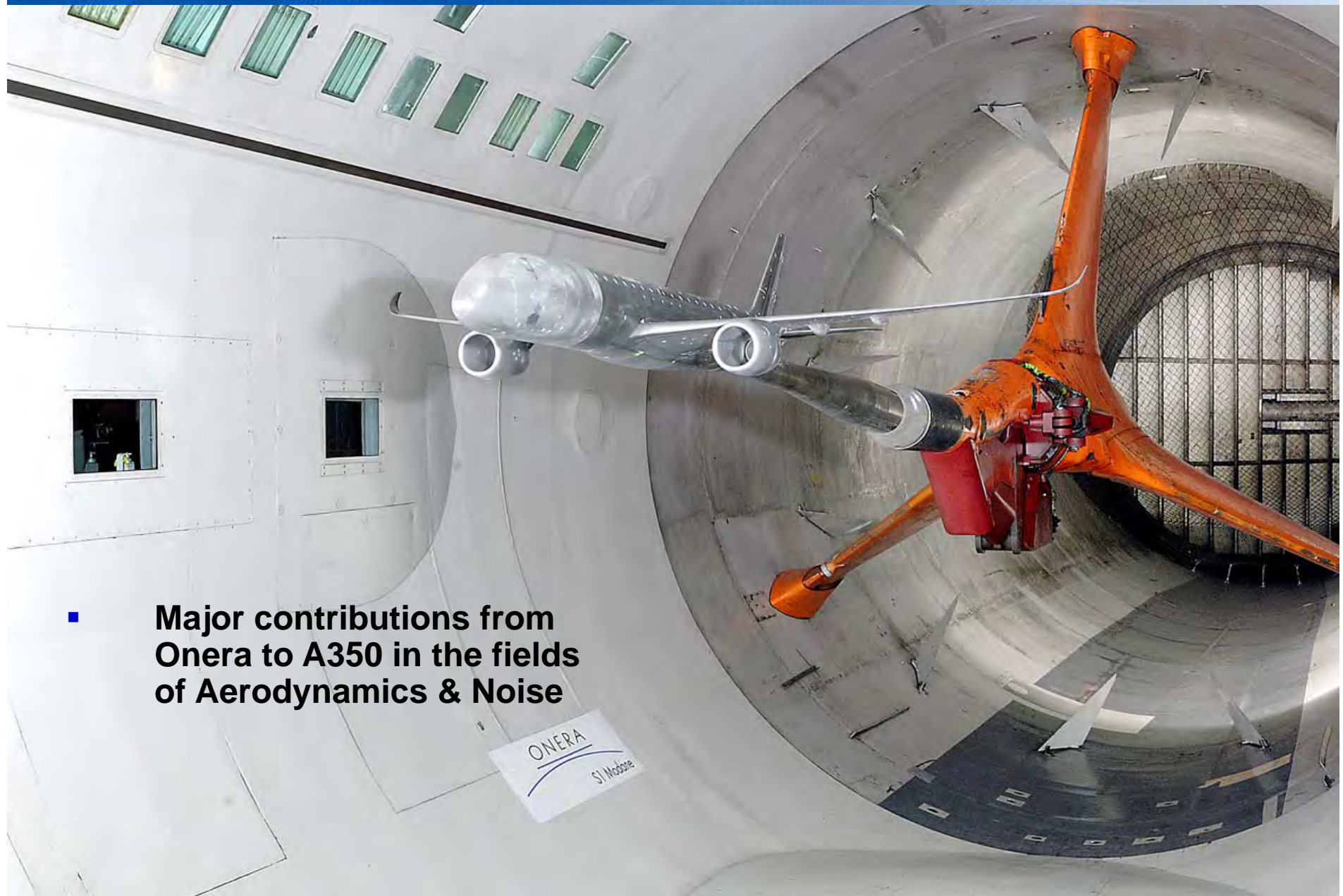
# A 380 large model in the Onera S1MA and F1 wind tunnels



- Major contributions from Onera to A380 in the fields of Aerodynamics & Noise



## A 350 large model in the Onera-S1MA wind tunnel

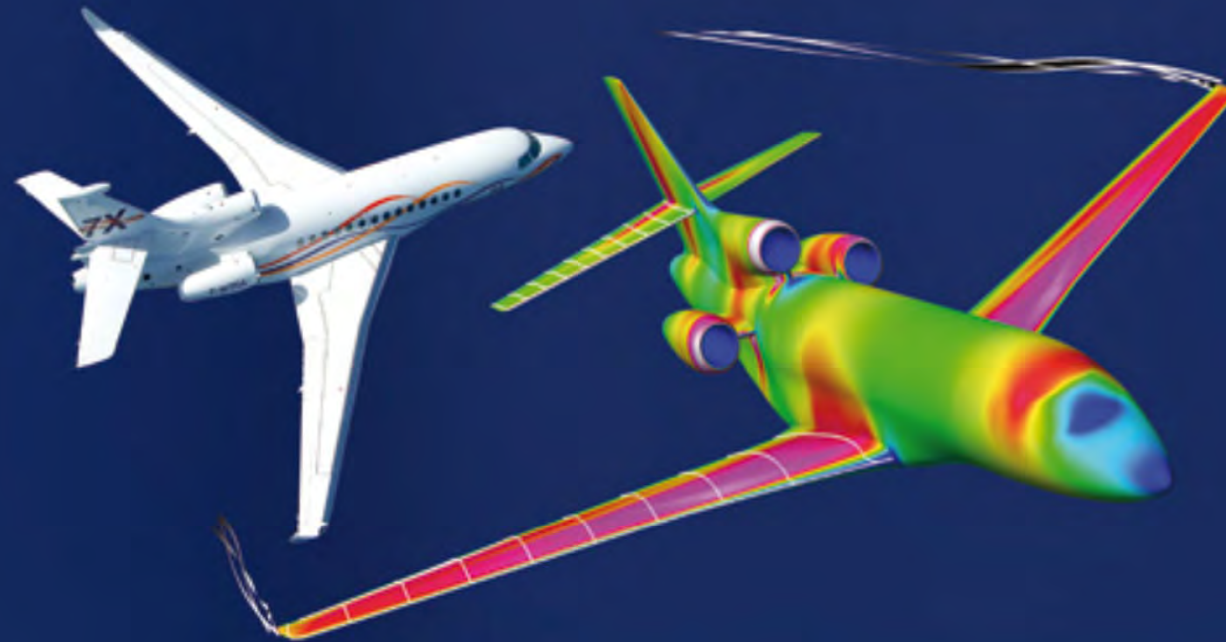


- Major contributions from Onera to A350 in the fields of Aerodynamics & Noise

# Winglets

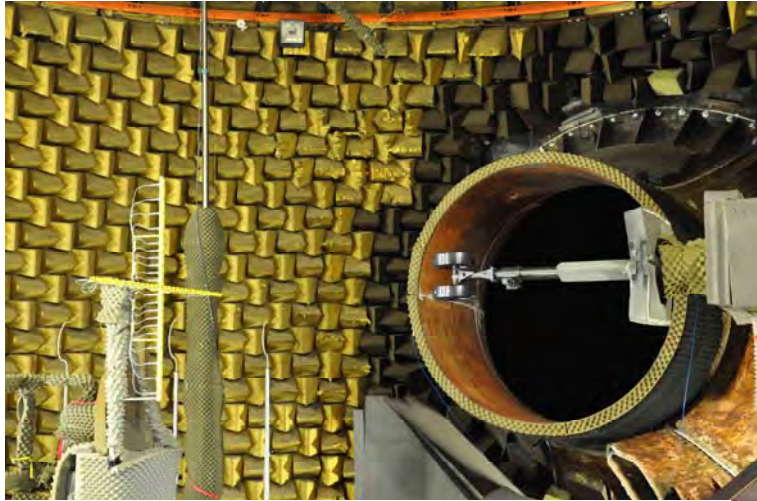
**An additional « device »,  
with a high potential for  
performance  
improvement of the  
aircraft (consumption,  
range)**

**Wind tunnels were  
major contributors to  
the assessment**

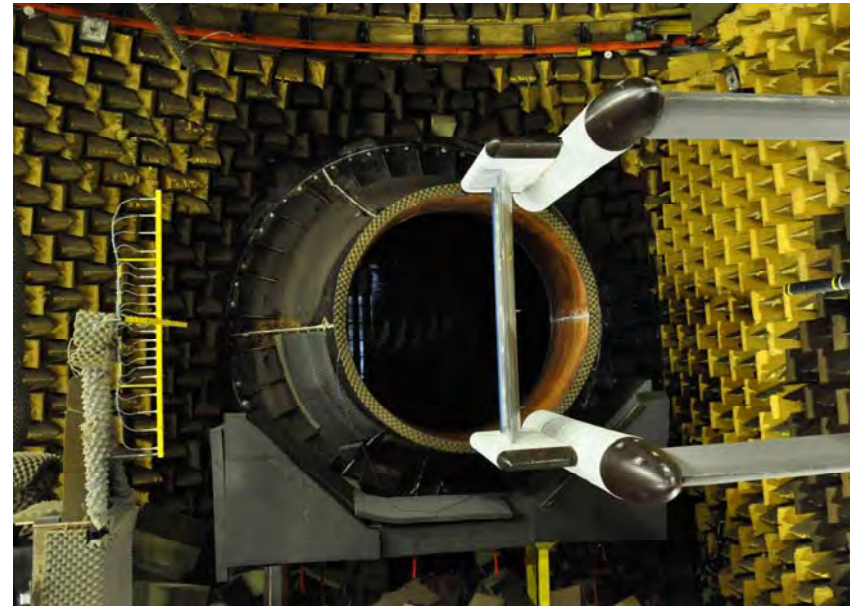




# Illustrations of recent activities in CEPRA19 : Aerodynamic Noise



**LAGOON (Airbus - DLR - ONERA)**  
Generic undercarriage leg

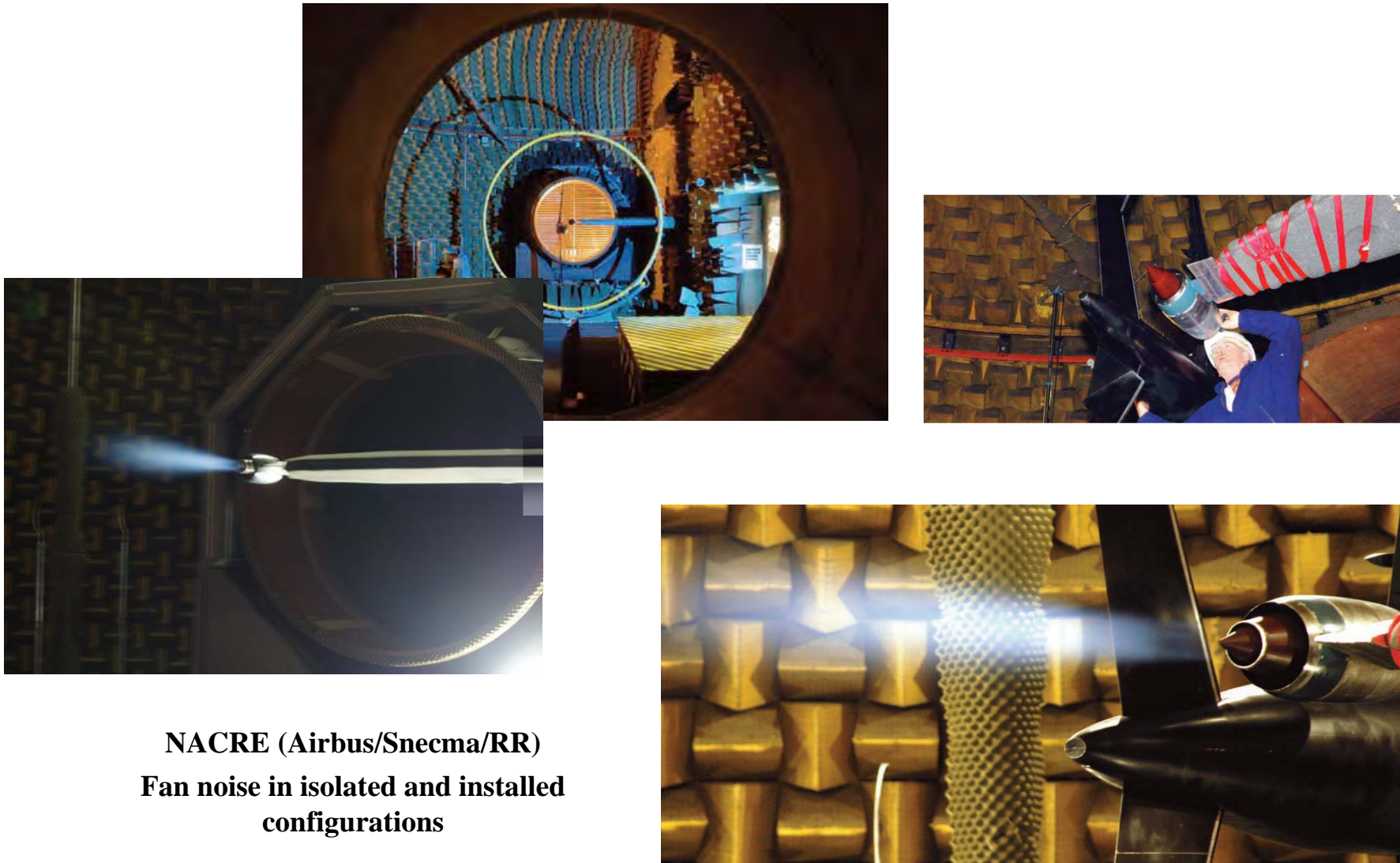


**TIMPAN (Airbus)**  
Wing section with High-Lift Devices  
and mesh across the slat gap



**DTP BAF (Dassault)**  
Business jet

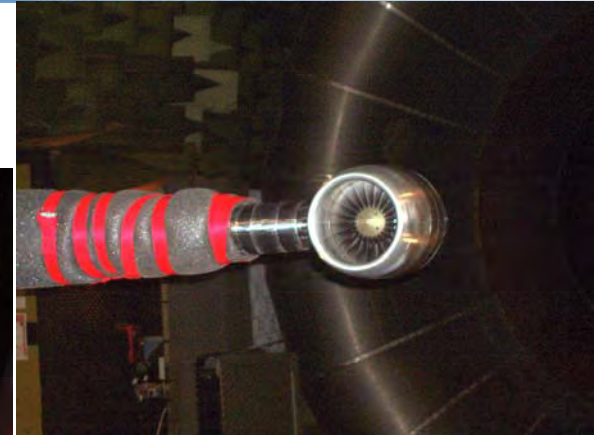
# Illustrations of recent activities in CEPRA19 : Fan Noise



**NACRE (Airbus/Snecma/RR)  
Fan noise in isolated and installed  
configurations**



## CEPRA19 – recent activities



- **Nacre : look on the configurations**

## elsA as a « Key Green Enabler »

**elsA is the ONERA software for complex external and internal flow simulations and for multi-disciplinary applications involving aerodynamics. That includes the following disciplines or topics:**

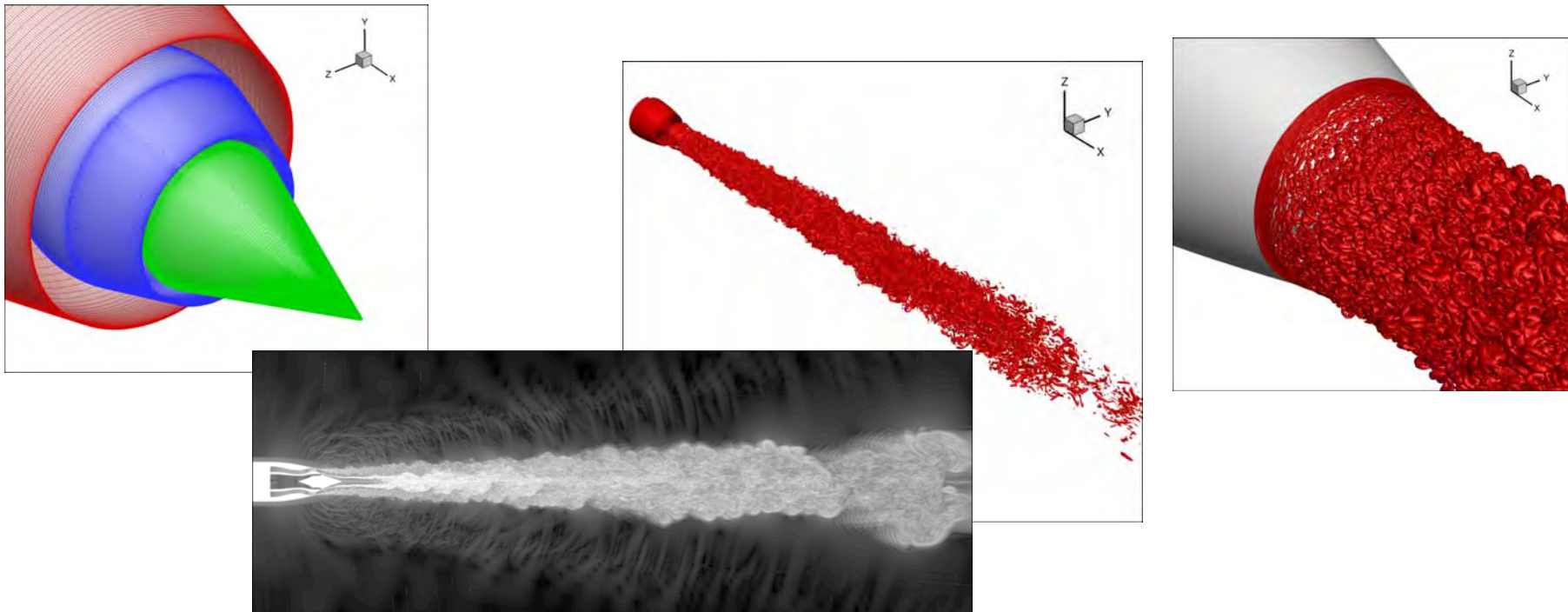
- **Aerodynamics, aeroelasticity, aerothermics coupling, aeroacoustics coupling;**
- **Aircrafts, helicopters, turbomachinery, missiles, launchers, air intakes, nozzles, propulsive jets;**
- **Research and industrial applications;**
- **Euler, RANS, URANS, DES, LES simulations;**
- **Mono-species perfect gas assumption with a user-given value of the specific heat ratio (equilibrium real gas assumption in development);**
- **Calculation of sensitivities for optimum design.**

**The objectives of the elsA project aim at providing a software that is : a federative support for research on physical and numerical modelling; RANS, URANS, DES, LES simulations;**

- **a long-term facility for capitalizing research results;**
- **Calculation of sensitivities**
- **a tool helping to investigate and understand flow physics;**
- **a powerful and user-friendly multi-purpose tool for applied CFD and multi-physics;**
- **a medium from research to industry;**
- **a tool open to co-operative work.**



# elsA for aérodynamique of jets

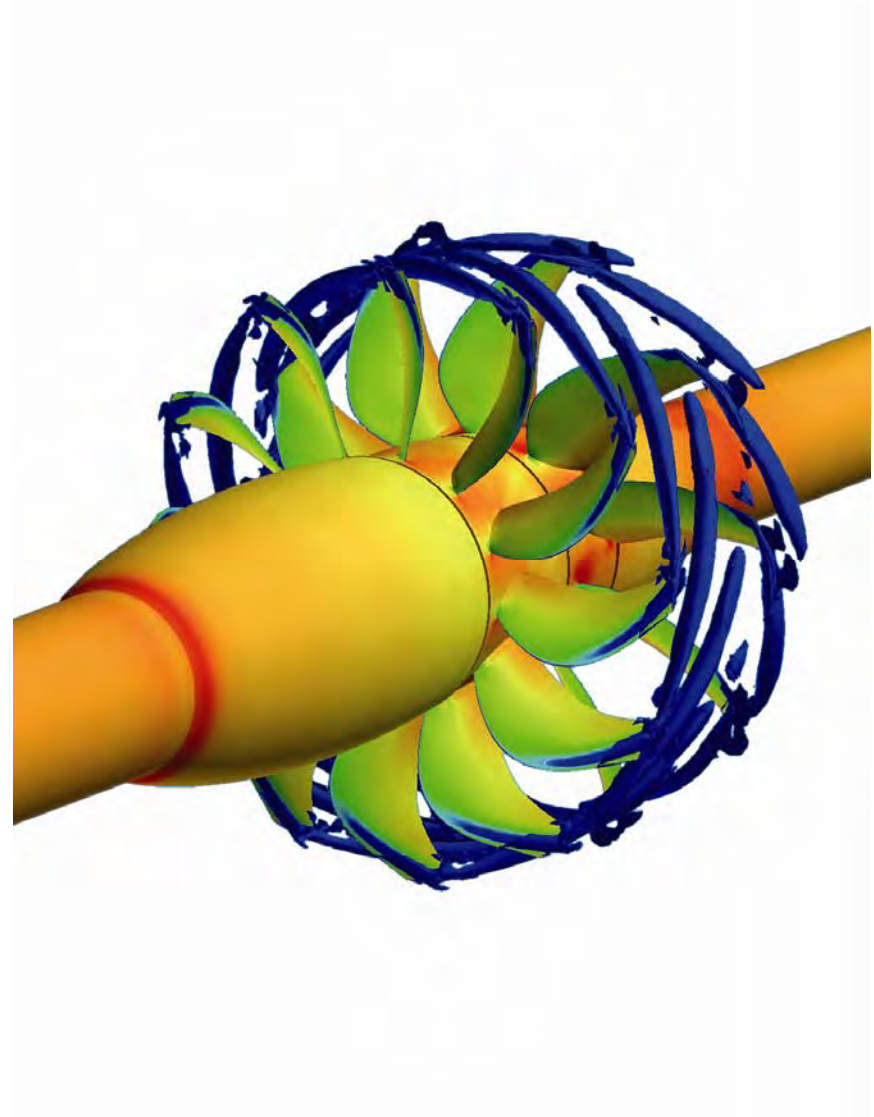


- ZDES computation
- 290 millions pts
- 500 000 h CPU

# CROR in Onera : a long story

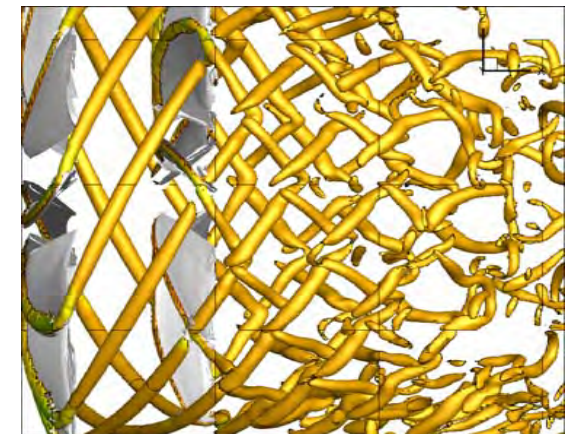
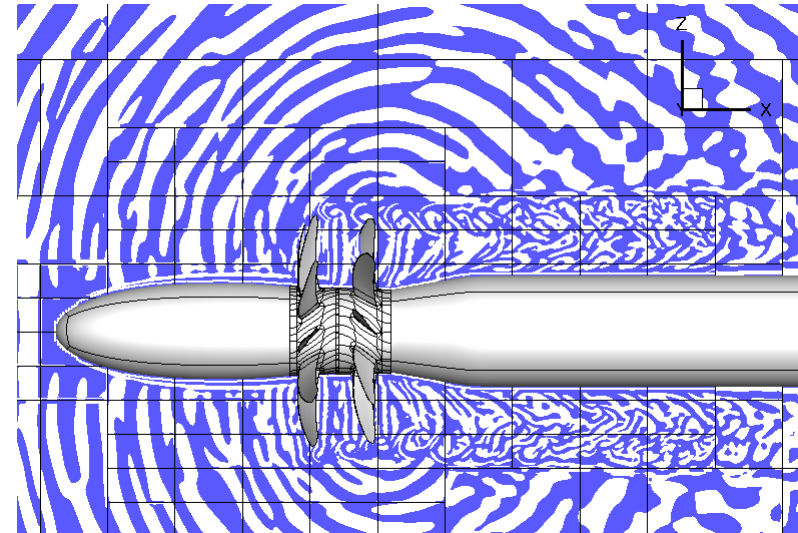
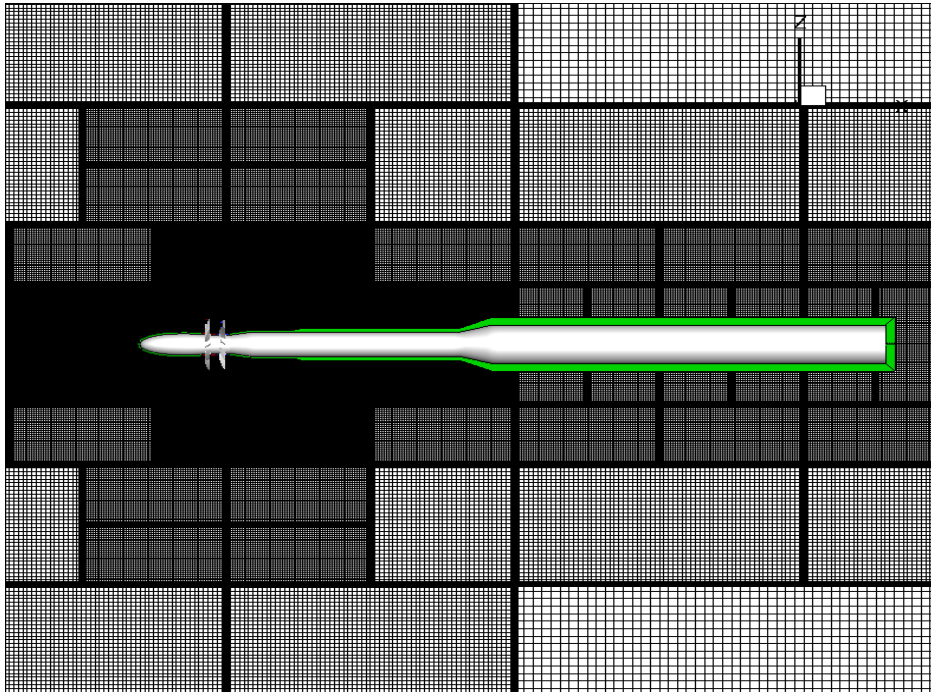
**Strong expertise in that field**

- **comparison simulations / tests**
- **interactions wake/rotor**
- **CROR demonstrators in windtunnel S1 Modane**





# elsA for CROR

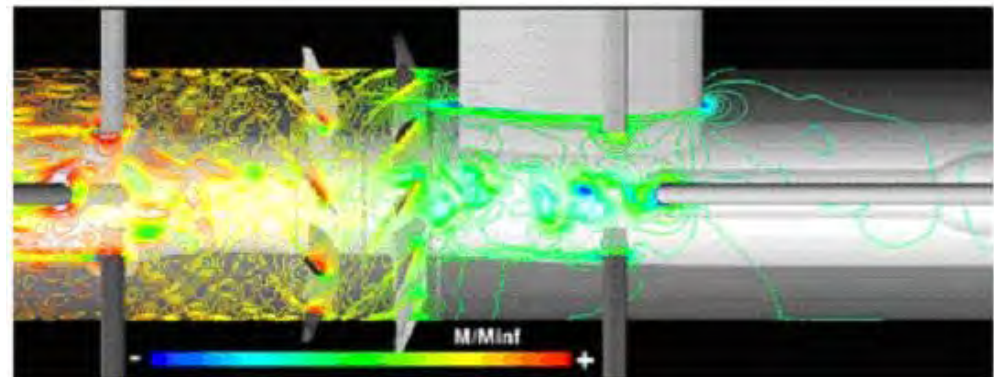
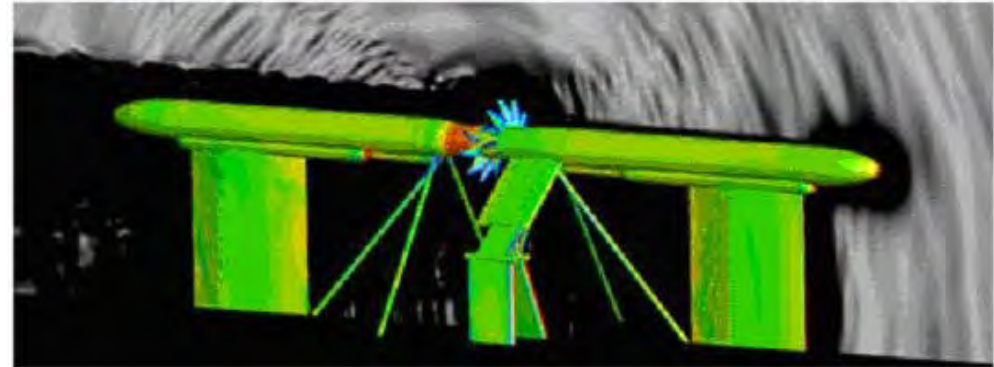


- elsA computation, approach « Chimère-Cartesian »
- 138 millions pts +480 millions pts
- 250 000 h CPU
- Following studies with pylon

## CROR : Test vs simulation

### Comparison between tests and CFD

- match Onera (URANS) TsAGI (tests)
- Far field acoustic visualisation
- 2 elsA-based computations on Stelvio (Onera supercomputer)
- 220 millions pts + Chimère technics
- 600 000 h CPU
- 2 publications (AIAA + AAAF 2012).





# CROR : Pylon

## Aerodynamics – interaction wake/OR

- key point for noise
- URANS technics needed
- comparison of different solutions
- 42nd AIAA Flow Control Conference : "Blowing Strategies of Pylon-Propeller Configuration for Noise Reduction Using Numerical Approach" 25-28 juin 2012, New-Orleans.
- further studies on new concepts for aeroacoustics

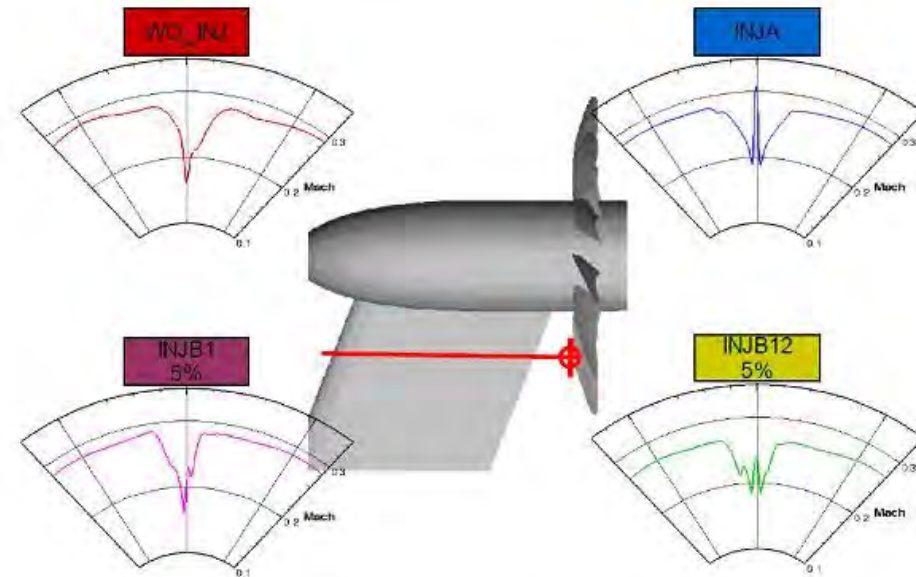
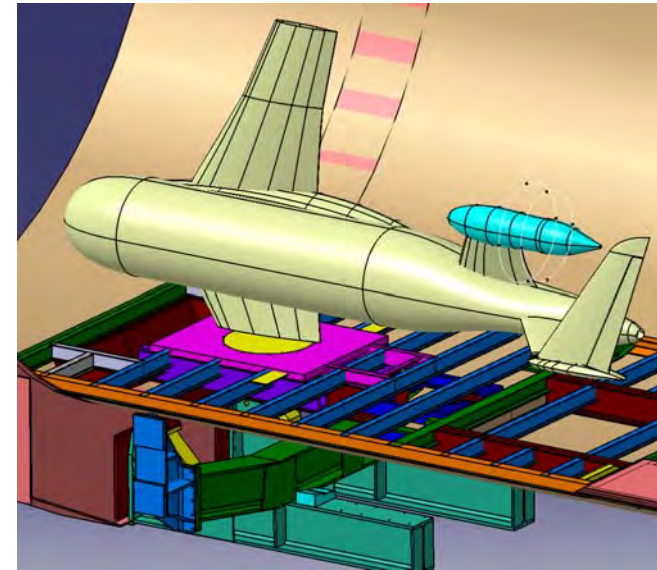
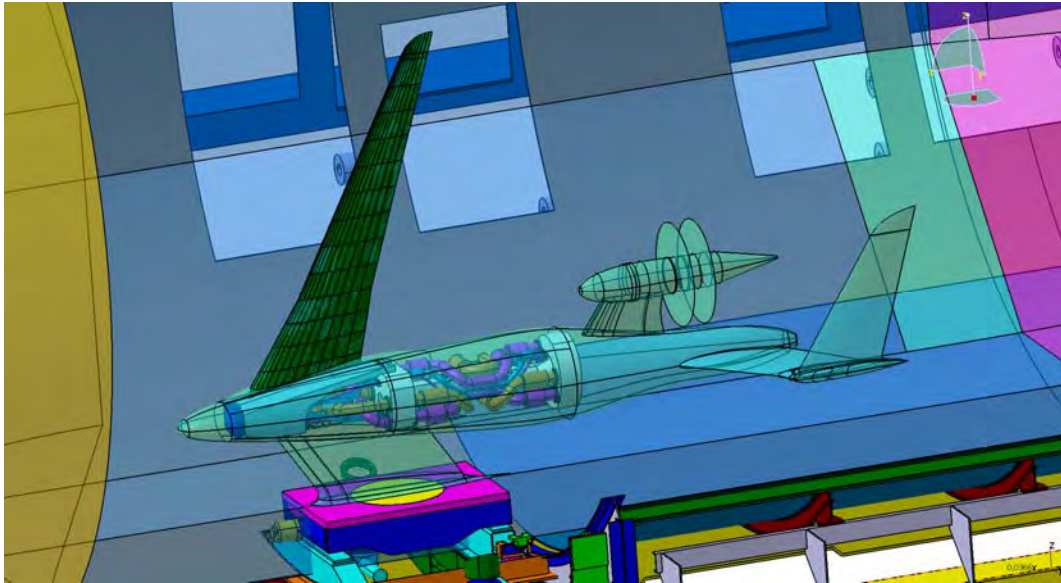
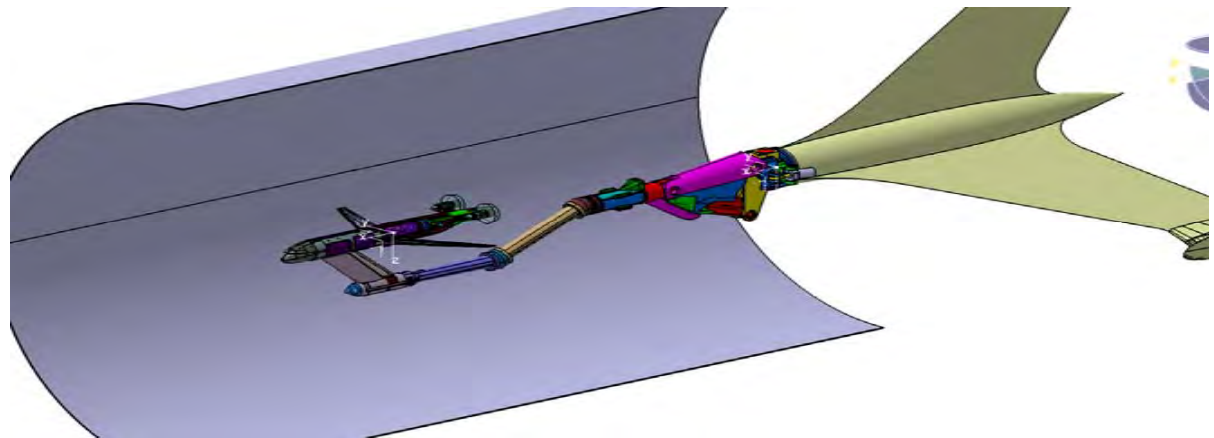


Fig 7. Wake visualization between the pylon trailing edge and the propeller leading edge at  $r/R = 50\%$ .

## Onera WTs to support development of CROR concept



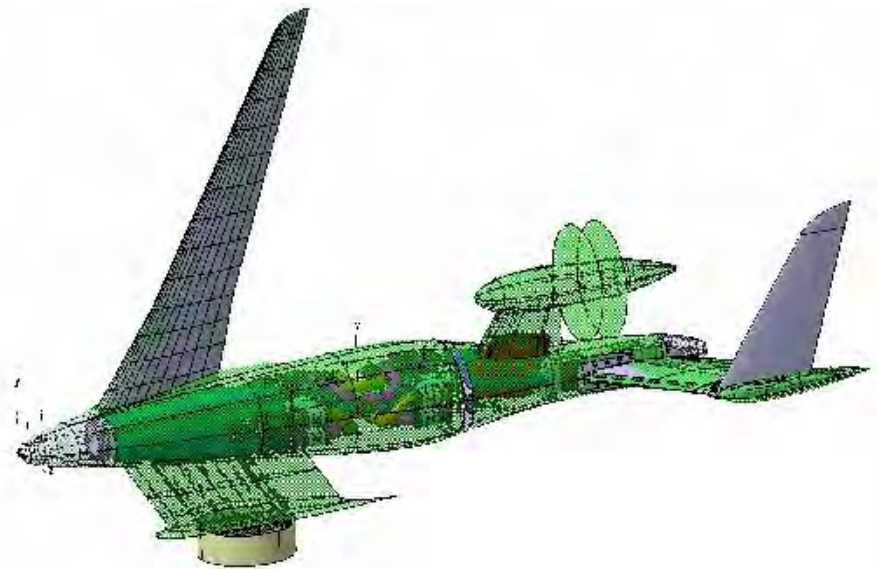
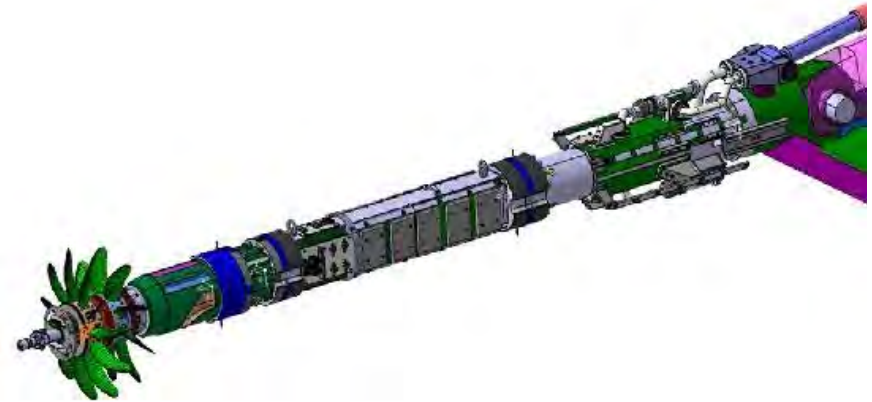
Test rigs are being developed for experiments in S1MA : objectives are performances and acoustics





## Z49, scale 1/5

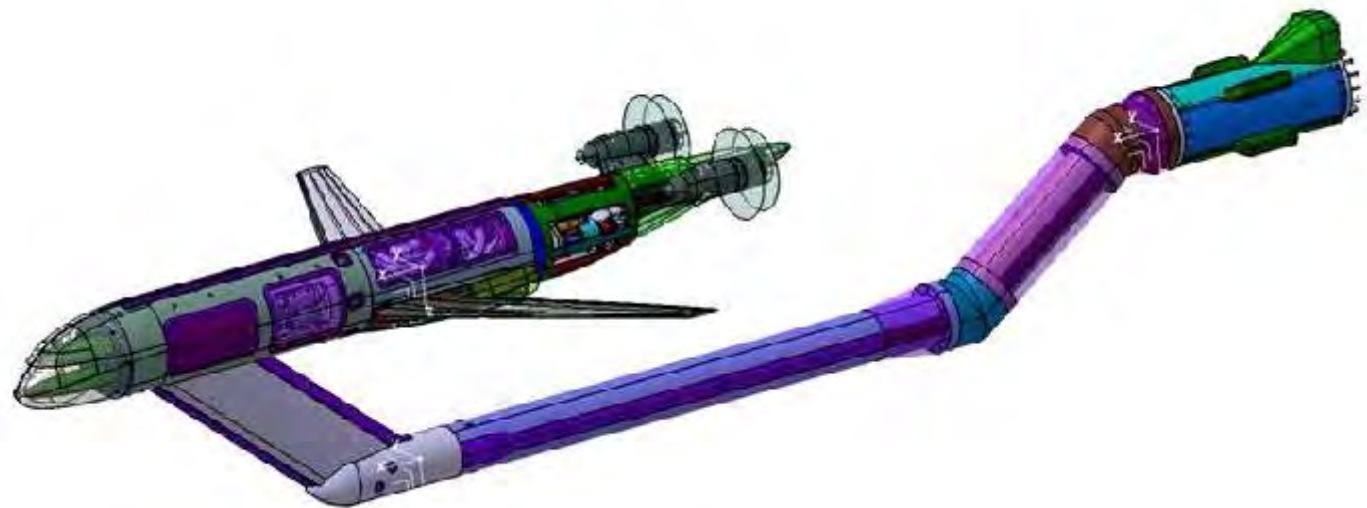
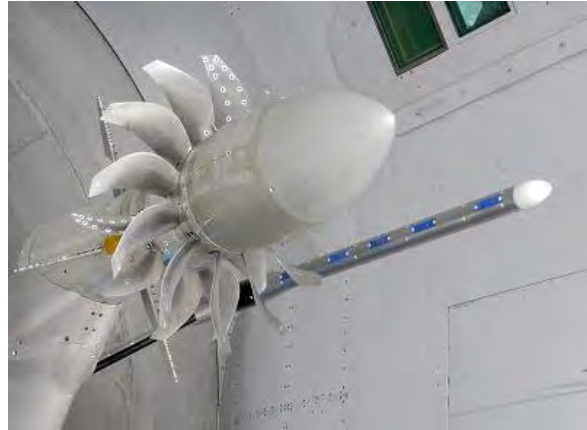
- Design : Onera
- Configuration Z49 alone
- Configuration Z49 with installation
- Specific technics (made in Onera) for blade deformations



# CROR & experimentations

Onera as a key player :

- HERA in S1MA
- Z10 scale 1/10





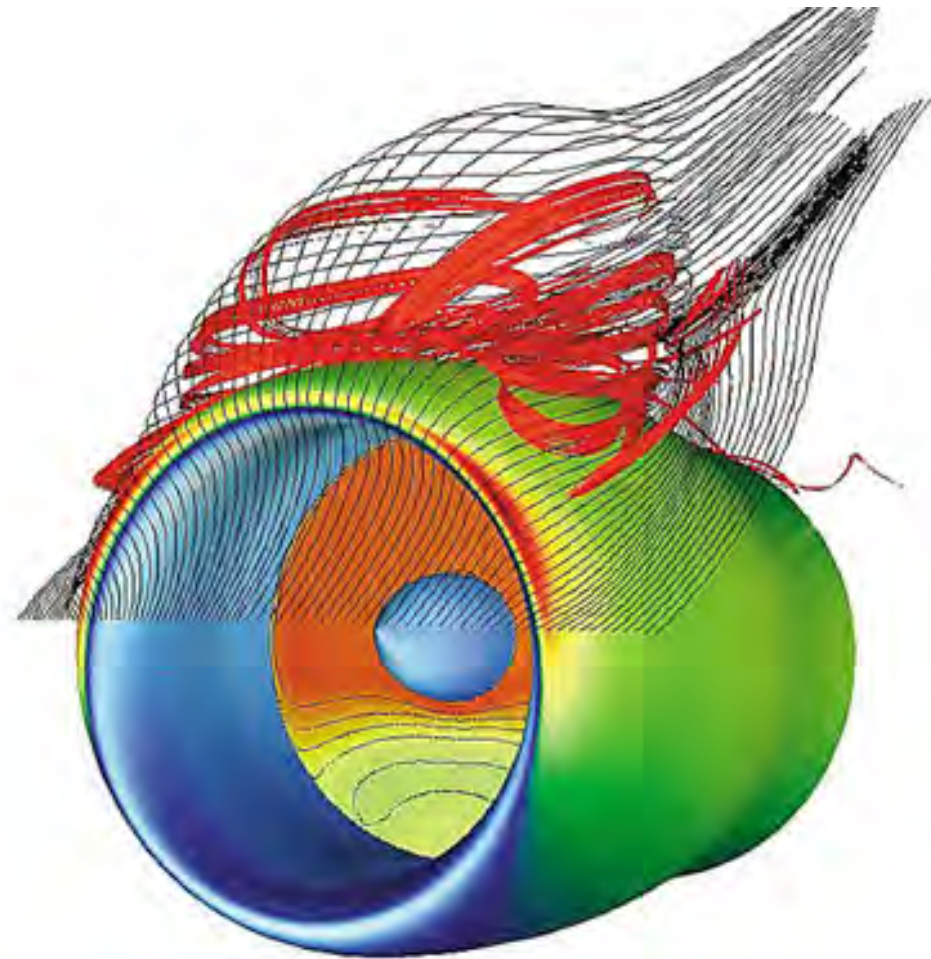
# Open-Rotor testing in the Onera S1MA large transonic wind tunnel



# challenges

## engine to wing and fuselage integration

- drag reduction
- BD2 testbed for tuyeres,
- S4B testbed for TPS & fuselage integration
- New tools for design





# Challenges

natural or hybrid laminar flow on wings :

- drag reduction
- old story but new ideas
- large model needed



# Onera : travelling together to our future – a lot to do !







Thank You !

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